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DEPARTMENT OF MECHANICAL ENGINEERING
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On the Control of Quadrupedal Bounding with a Flexible Torso and a Tail

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Presentation Overview

- Introduction – Quadrupeds with Flexible Torso & Tail
- Learning from Biology
- Attitude Control
- Coronal Plane Dynamics - Simulation
- Sagittal Plane Dynamics - Simulation
- 3D Model Attitude Control
- Simulation Results
- Conclusions

Introduction - Legged Robots

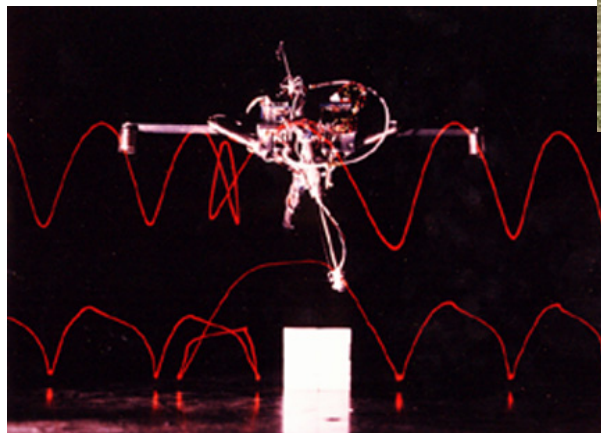
- Point contact with ground
- Move on terrain with discontinuities
- Obstacle avoidance



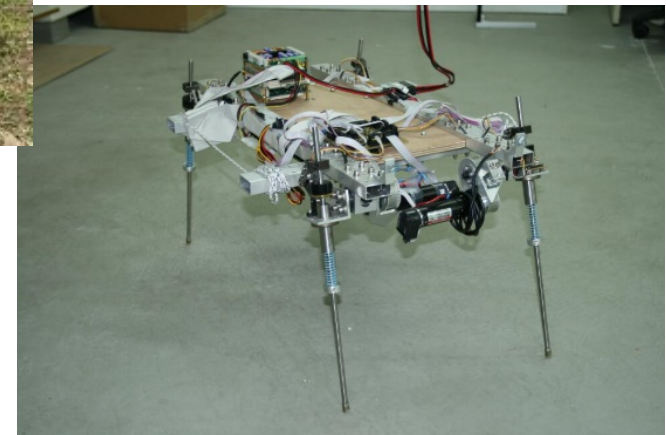
BigDog



RHex



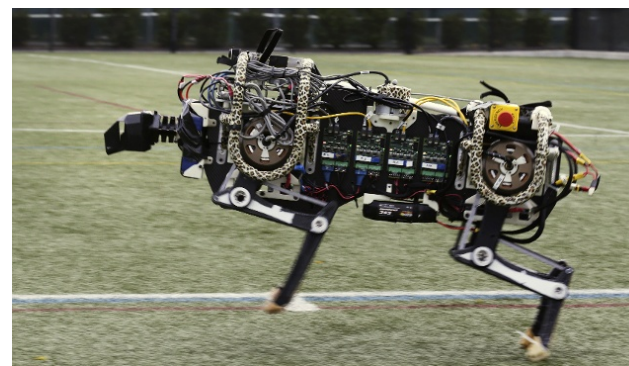
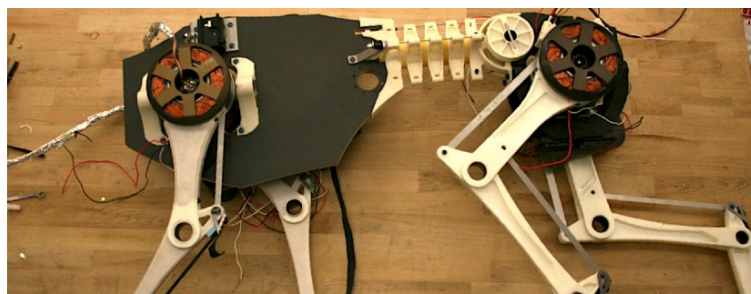
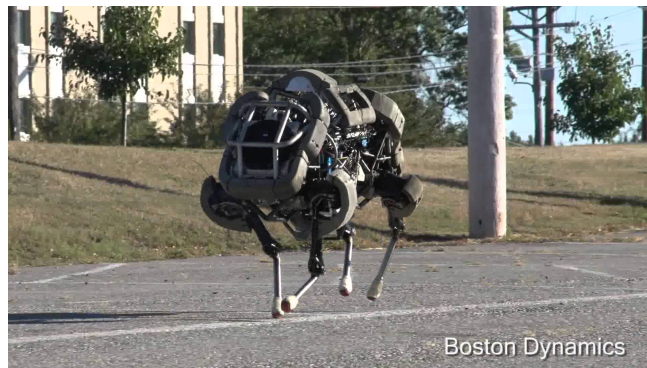
Raibert's monopod



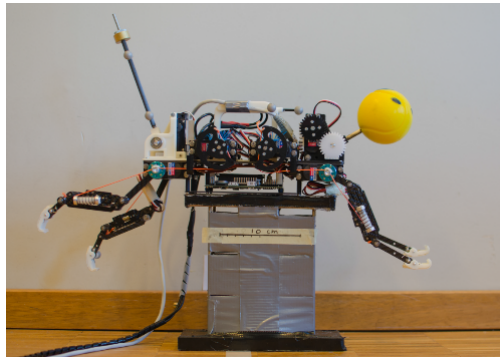
NTUA's Quadruped

Introduction - High Performance Quadrupeds

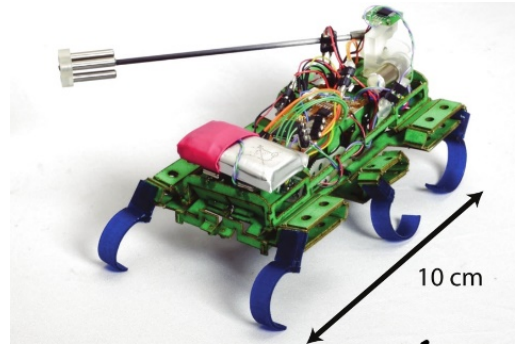
- Boston Dynamics: Cheetah (13 m/s) - Wildcat
- MIT: Cheetah (6 m/s)
- Slower than their natural counterpart (29 m/s = 104 km/h)



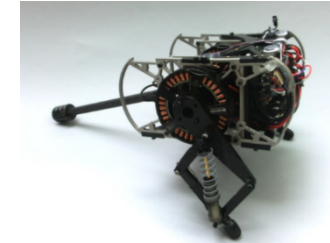
Introduction - Robots with Tails



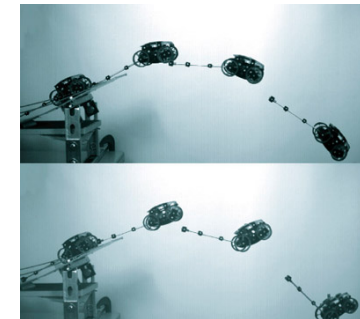
Cheetah-Cub
EPFL - Pitch



TAYLRoACH, University of
California, Berkeley - Yaw



Penn Jerboa, Upenn - Pitch



Tailbot, University of California,
Berkeley - Pitch



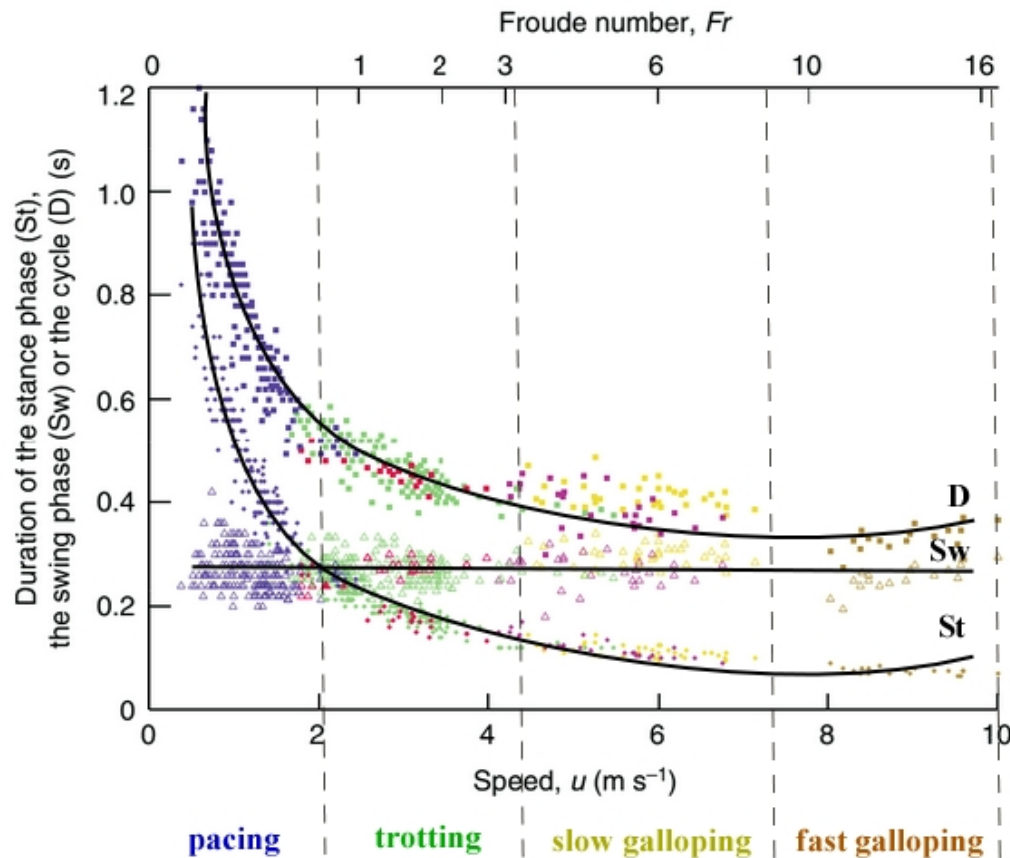
XRL, Upenn - Pitch



FlipBot, University of
Cape Town - Roll, Pitch

- Numerous legged robots have been designed, but only a **minority** of them **include appendages** for angular momentum management, such as tails, or reaction wheels.

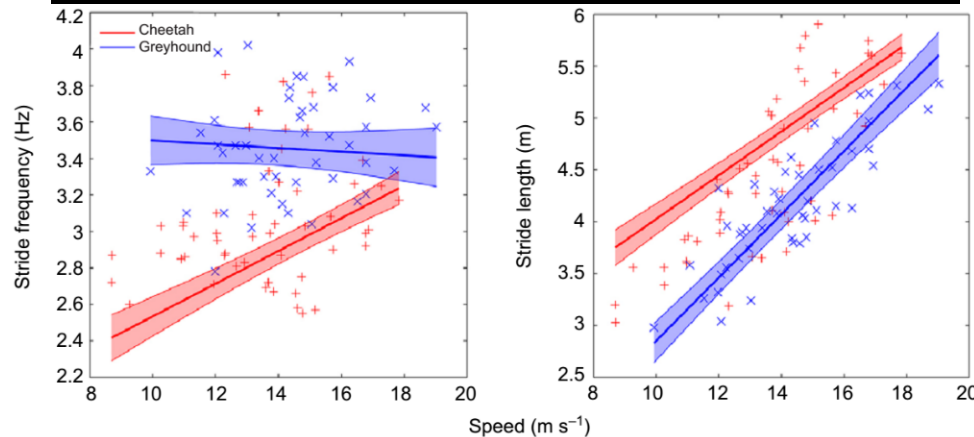
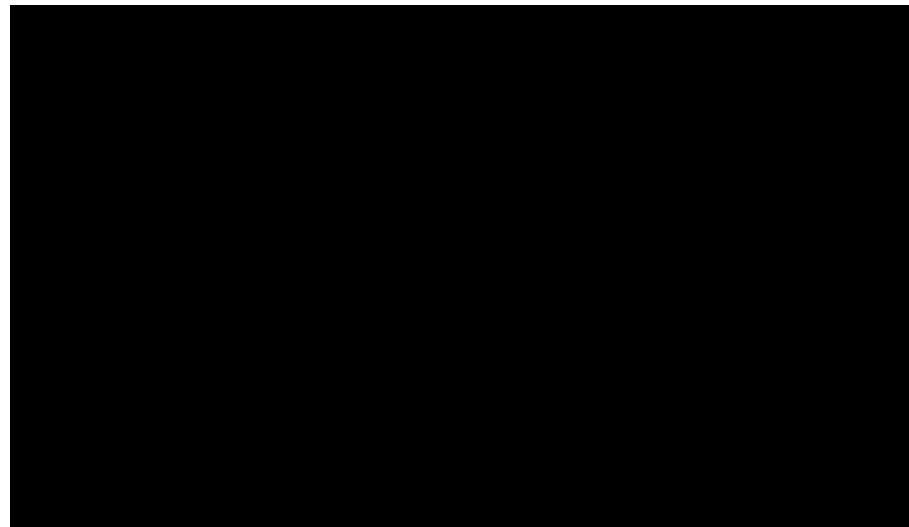
Learning from Biology - Increasing Speed



Gait duration and speed

- Data from Belgian Shepards
- Stance duration (St) drops with speed
- Swing duration (Sw) constant
- Over 5 m/s, stride duration (D) cannot be reduced
- High speed galloping ($v > 8 \text{ m/s}$) achieved with extensive torso/spine motion

Learning from Biology – Flexible Torso



Stride frequency

Stride length

Flexible torso benefits

- Increased **stride length** by placing legs forward
- Energy storage at spine muscles tendons => **increased energy efficiency**
- **Low stride frequency**
 - the cheetah reaches the same forward velocity with lower frequency motion than the racing greyhound
- **Stability of motion**

Learning from Biology - Animal Tails

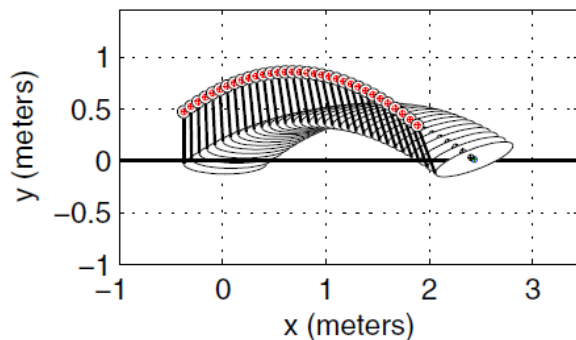


- Many legged animals have long tails which aid in balance and maneuverability at high speeds.
- Tail motion effective for adjustments to unexpected perturbations, when the legs are otherwise occupied.

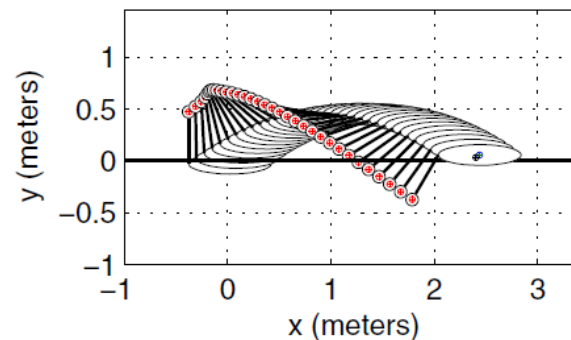


Attitude Control

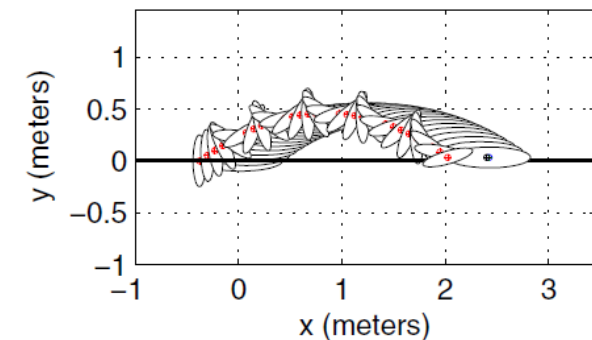
- Quadruped robots are *highly underactuated* machines.
- Tasks such as high speed galloping, jumping over obstacles, or gait transitions, require *precise control* of their *attitude*.
- Attitude control is *achieved indirectly* through the *motion of the legs*
 - This technique assigns more control tasks to the legs forcing them to trade-offs that may lead to low performance.
- To mitigate this challenge, *dedicated appendages* with large moment of inertia (Mol) can be used.



Unacceptable

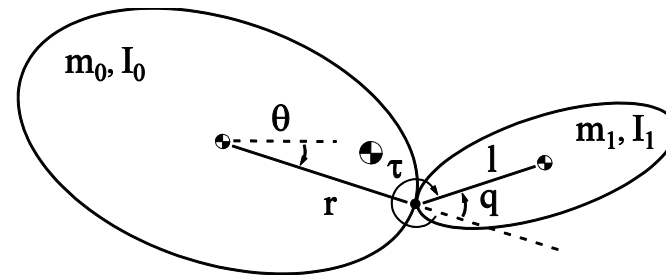
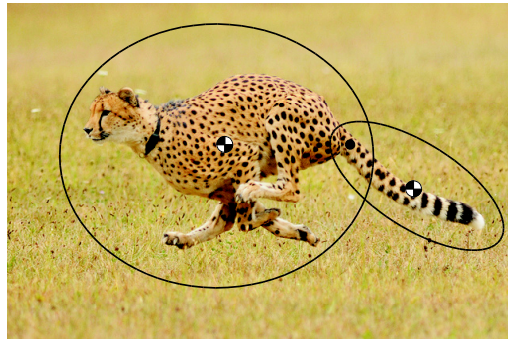


Tail



Wheel

General 2D Dynamic Model in Aerial Phase



- Equations of Motion (EoM) in Aerial Phase

$$(I_0 + I_1 + \mu(l^2 + r^2 + 2rl \cos q))\ddot{\theta} + (I_1 + \mu(l^2 + rl \cos q))\ddot{q} - \mu rl \sin q \dot{q}^2 - 2\mu rl \sin q \dot{q} \dot{\theta} = 0$$

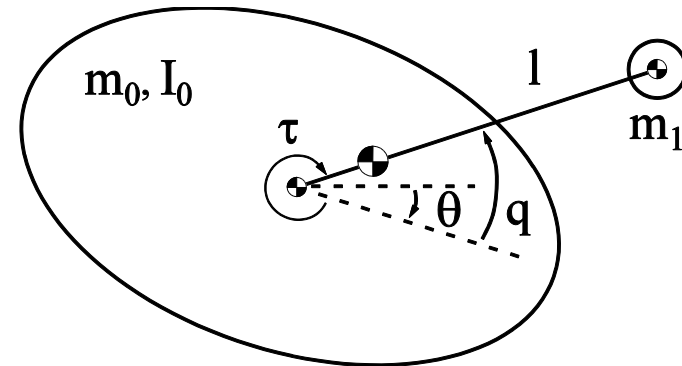
$$(I_1 + \mu l^2 + \mu rl \cos q)\ddot{\theta} + (I_1 + \mu l^2)\ddot{q} + \mu rl \sin q \dot{\theta}^2 = \tau$$

- Conservation of Angular Momentum

$$(I_0 + I_1 + \mu(l^2 + r^2 + 2rl \cos q))\dot{\theta} + (I_1 + \mu(l^2 + rl \cos q))\dot{q} = h_0$$

- The analysis holds for body maneuvers in **roll, pitch, and yaw**, assuming these motions are decoupled.
- Next, we examine maneuvers on the **coronal plane** trying to control the **roll** angle of the body.

Dynamics and Control on the Coronal Plane



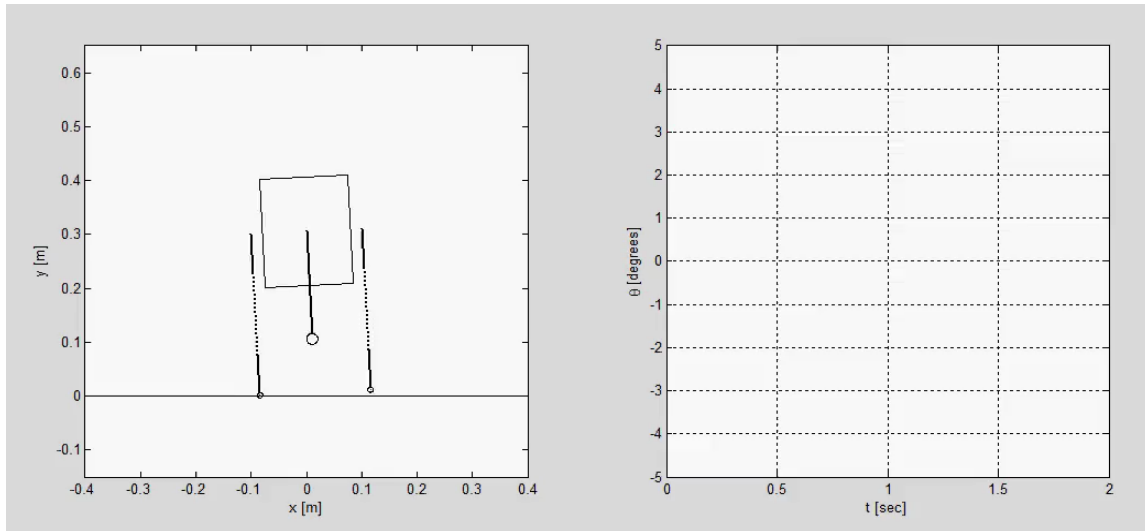
- For $I_1=0$, $r=0$, the EoM decouple, and the tail angle q becomes an *ignorable* coordinate.
- As observed in the first equation, we can control roll angle θ , with a simple controller of the form:
- The PD gains k_p , k_v depend on **the time available** to complete the maneuver, t_s .

$$I_0 \ddot{\theta} = -\tau, \quad \frac{I_0 \mu l^2}{I_0 + \mu l^2} \ddot{q} = \tau$$

$$\tau = -I_0 (k_v \dot{e}_\theta + k_p e_\theta)$$

$$k_p = 36/t_s^2, \quad k_v = 12/t_s$$

Adding Legs and Ground Forces to the Model



Parameters

Body: $I_0=0.3\text{kgm}^2$,
 $m_0=20.8\text{kg}$

Hip to Hip distance: 0.1m

Leg: $l=0.32\text{m}$, $k=3523\text{N/m}$

Tail: $m_1=0.5\text{kg}$, $l=0.2\text{m}$.

Initial Conditions

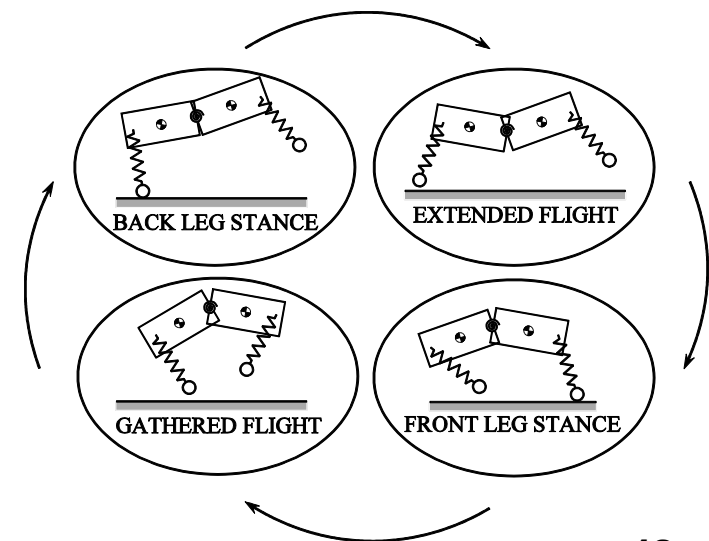
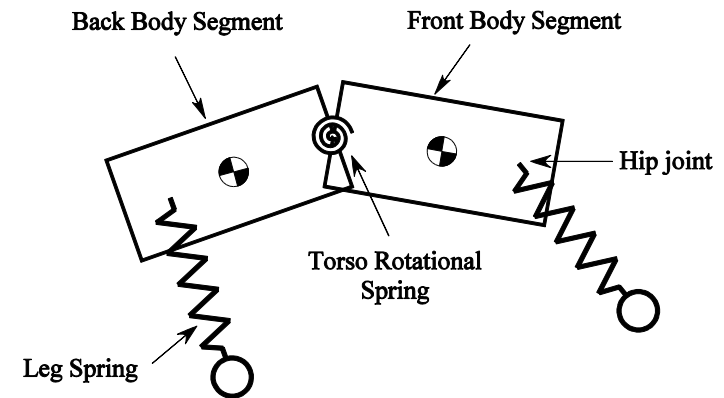
$\theta_0=3^\circ$, $y_0=0.35\text{m}$

- Initial conditions and parameters obtained from a 3D experiment in ADAMS that started with *roll angle* $\theta_0=3^\circ$.
- The controller designed previously is used in aerial phases to bring the *body angle to zero*. The tail manages to *stabilize* the body angle.

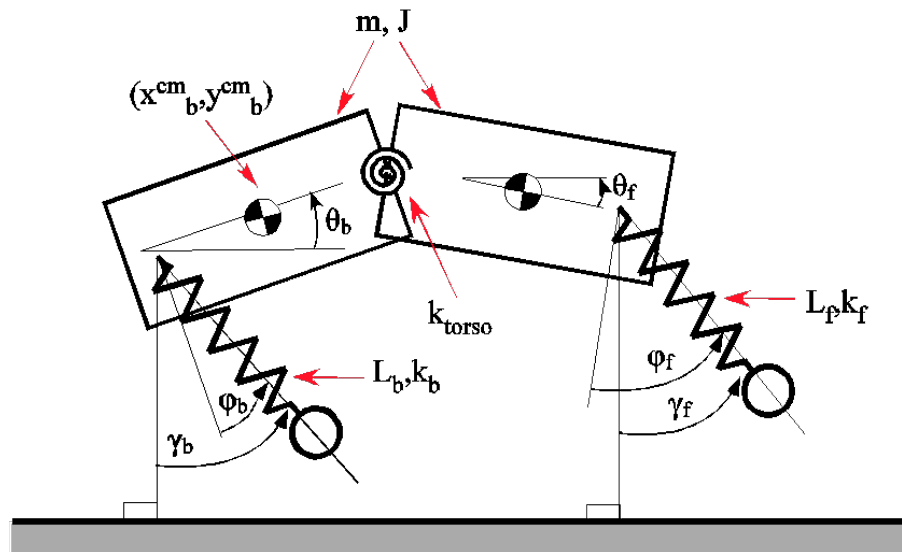
Modeling and Dynamics: Sagittal Plane

Assumptions & simplifications

- Motion at sagittal plane (2D) only
- Simplified galloping = bounding
- Two – segment body
- Pinned spine joint with torsional spring
- Massless springy legs
- No sliding during foot-ground contact
- Passive model



Modeling and Dynamics: Double Stance



Equations derived with
Lagrange's methodology

Generalized coordinates

$x_b^{cm}, y_b^{cm}, \theta_b, \theta_f$

Control inputs

$\varphi_b^{td}, \varphi_f^{td}$

$$2m\ddot{x}_b^{cm} + m(-d\dot{\theta}_b^2 \cos\theta_b - d\dot{\theta}_f^2 \cos\theta_f - d\dot{\theta}_b^2 \sin\theta_b - d\dot{\theta}_f^2 \sin\theta_f) + k_b(L_b - l_b)\sin\gamma_b + k_f(L_f - l_f)\sin\gamma_f = 0$$

$$2m\ddot{y}_b^{cm} + m(-d\dot{\theta}_b^2 \sin\theta_b - d\dot{\theta}_f^2 \sin\theta_f + d\dot{\theta}_b^2 \cos\theta_b - d\dot{\theta}_f^2 \cos\theta_f) - k_b(L_b - l_b)\cos\gamma_b - k_f(L_f - l_f)\cos\gamma_f + 2mg = 0$$

$$\ddot{\theta}_b(J + md^2) + \ddot{\theta}_f md^2 - \ddot{x}_b^{cm} d \sin\theta_b + \ddot{y}_b^{cm} d \cos\theta_b - dk_b(L_b - l_b)\cos(\gamma_f - \theta_b) - dk_f(L_f - l_f)\cos(\gamma_b - \theta_b) + k_t(\theta_b - \theta_f) + mgd \cos\theta_b = 0$$

$$\ddot{\theta}_f(J + md^2) + \ddot{\theta}_b md^2 - \ddot{x}_b^{cm} d \sin\theta_f + \ddot{y}_b^{cm} d \cos\theta_f - 2dk_f(L_f - l_f)\cos(\gamma_f - \theta_f) - k_t(\theta_b - \theta_f) + mgd \cos\theta_f = 0$$



Modeling and Dynamics: Poincaré Map

Poincaré section

Taken at the extended flight phase, at the apex of the spinal joint

Fixed points

Proper initial conditions + “control” inputs = continuous motion

$$\mathbf{z}_f[k+1]=P(\mathbf{z}_f[k], \alpha_f[k])$$

$$\mathbf{z}_f=[y_b^{cm}, \theta_b, \theta_f, \dot{x}_b^{cm}, \dot{\theta}_b, \dot{\theta}_f]^T$$

$$\alpha_f=[\varphi_b^{td}, \varphi_b^{td}]^T$$

Simulation: Parameters & Initial Conditions

Initial conditions

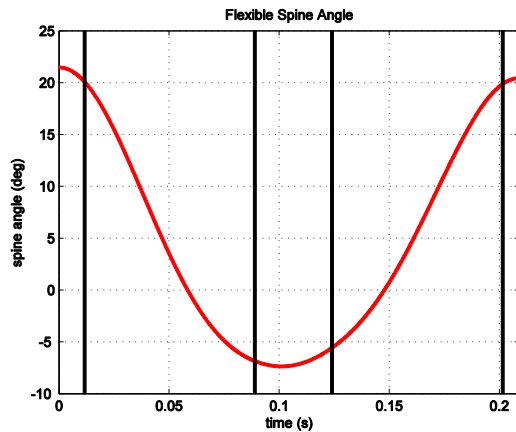
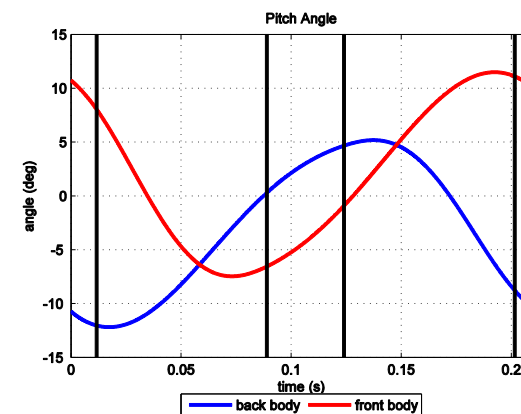
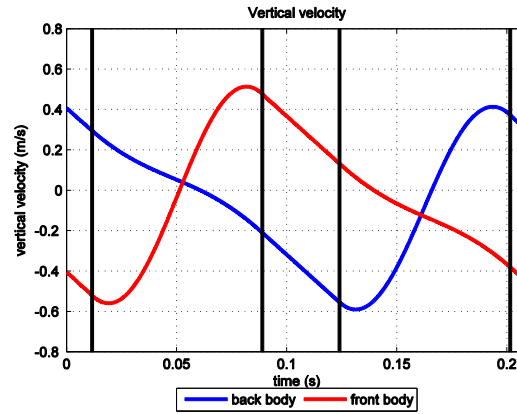
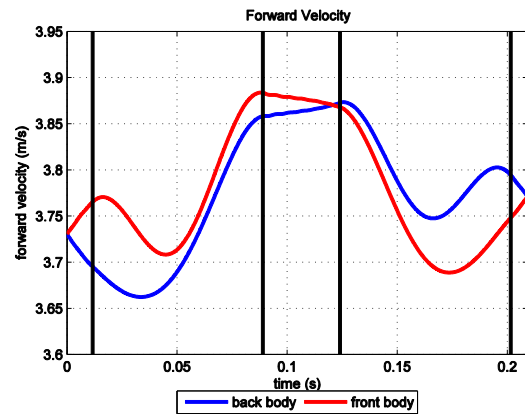
Parameter	Value
$\dot{x}_{b,0}^{cm}$	3.73 m/s
$\dot{y}_{b,0}^{cm}$	0.41 m/s
$y_{b,0}^{cm}$	0.29 m
$\theta_{b,0}$	-10.73 deg
$\theta_{f,0}$	10.73 deg
$\dot{\theta}_{b,0}$	-172 deg/s
$\dot{\theta}_{f,0}$	-172 deg/s

Robot parameters

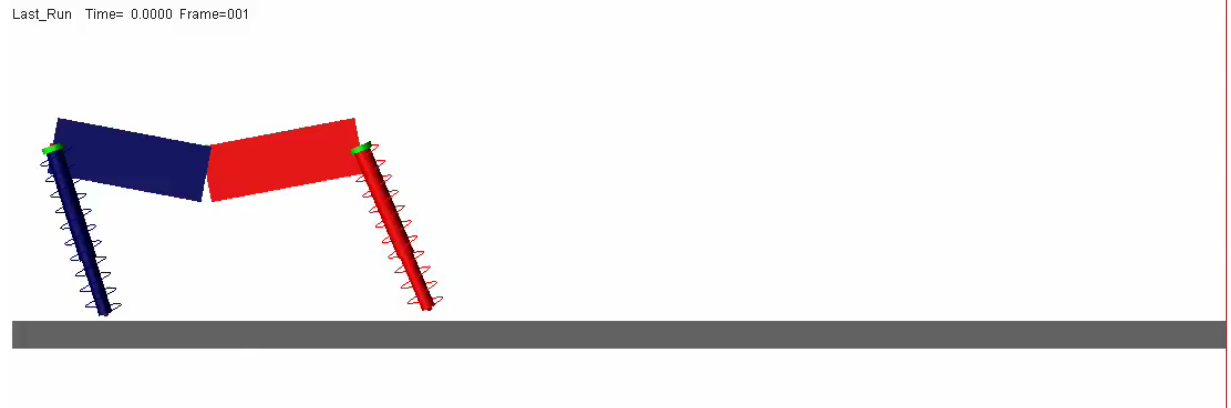
Parameter	Value
m	10.432 kg
J	0.339 kg m ²
$L_f = L_b$	0.323 m
d	0.138 m
$k_f = k_b$	7046 N/m
K_{torso}	165 N/m

Control input	Value
φ_b^{td}	28.14 deg
φ_f^{td}	12.12 deg

Simulation - Results



Last_Run Time= 0.0000 Frame=001



- Reversing time symmetry $\mathbf{x}_b(-t) = \mathbf{x}_f(t)$
 where $\mathbf{x}_i = [y_i^{cm}, \theta_i, \dot{\theta}_i, \dot{x}_i^{cm}, \dot{y}_i^{cm}]^T$, $i \in \{f, b\}$



3D Model with Tail & Flexible Torso

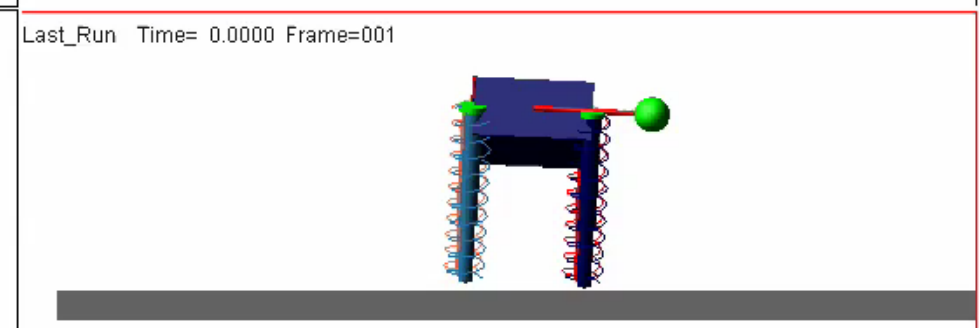
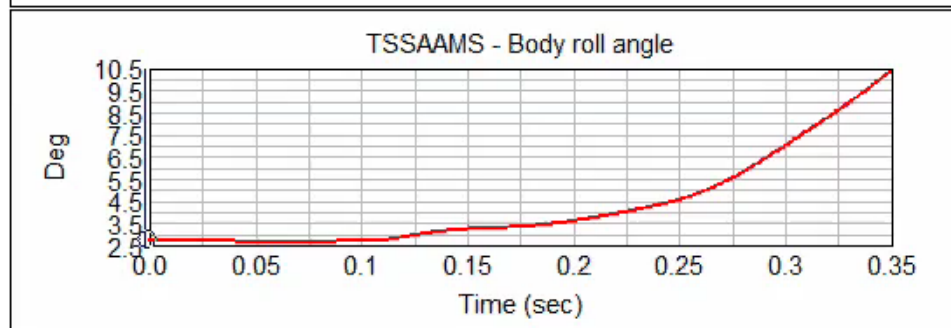
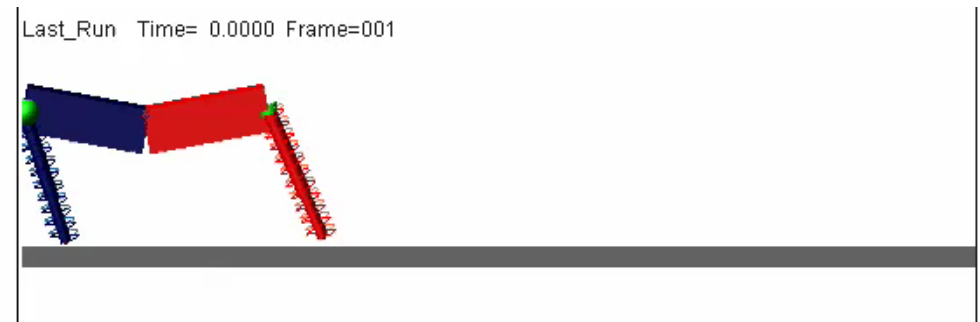
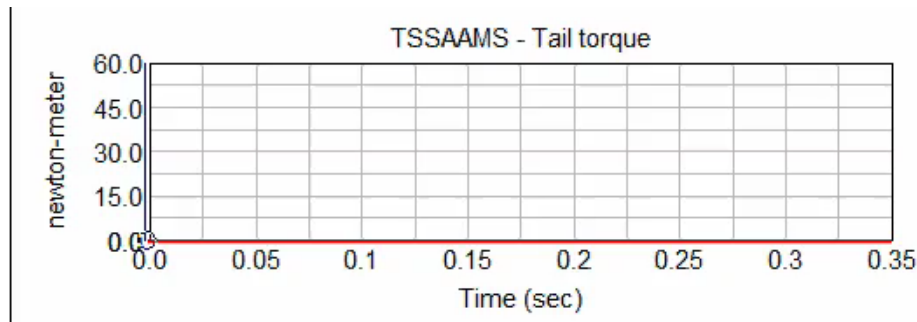
■ Model characteristics

- 3D model in MSC Adams simulation environment
- **Flexible torso** and **tail** in a single model
- **Initial conditions** and **leg touch down angles** derived from sagittal plane fixed point
- 3 deg initial roll
- **Tail active** only during **flight** – passive during stance
- Simple tail control law

■ Aim

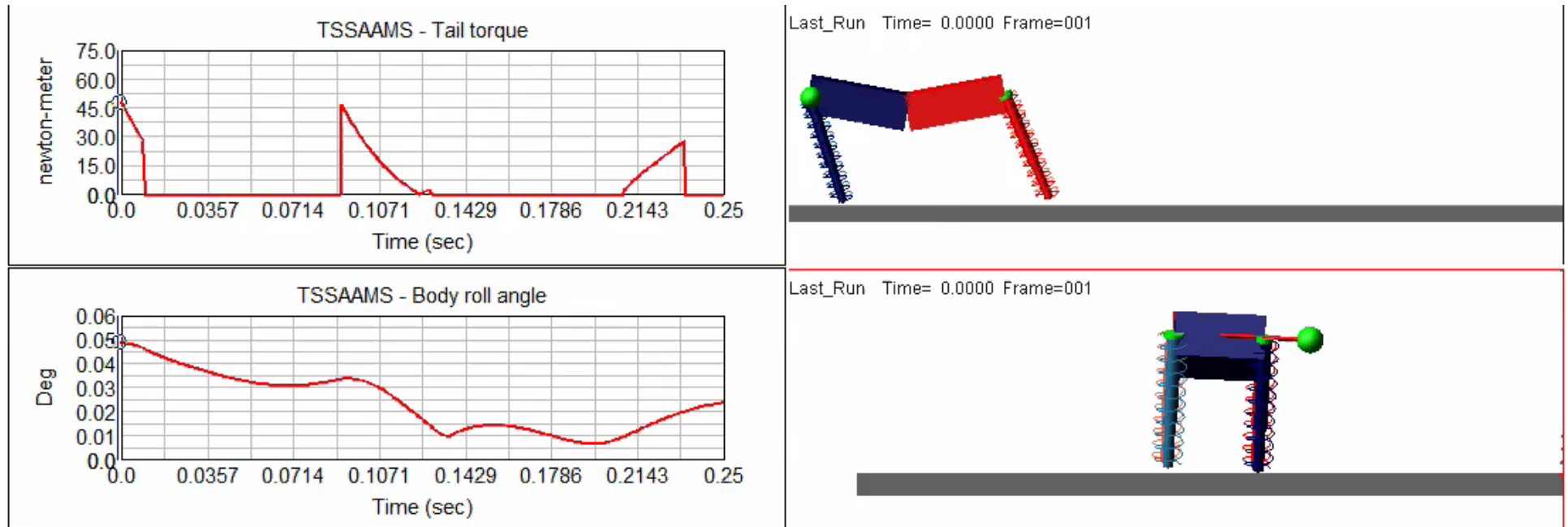
- To correct the initial roll angle and conserve the motion

Simulation – Passive Tail (no actuation)



- Roll angle increases since no control is applied
- Not enough foot clearance
- Incapable of completing one stride

Simulation – Success with Active Tail



- Tail actuation during flight phase only
- Roll angle correction within one stride
- Completed the stride *successfully*



Conclusions

- Derived the 2D passive dynamics of a quadruped with a *flexible torso*.
 - A *fixed point* of the motion was found.
- Derived the 2D model of a quadruped with a *tail-like appendage*.
 - A controller was developed to stabilize the main body attitude on the coronal plane using a *tail*.
- A complex 3D model with both a *flexible torso* and a *tail* was simulated with and without tail actuation on the coronal plane.
 - When starting with a roll angle of 3° , the robot *fails* to make a second stride without tail actuation.
 - With the tail correcting the roll angle of the body during flight, the 3D robot follows a *stable* motion corresponding to the fixed point found in the 2D analysis.

Acknowledgement

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